

MODELS: Wright Double Row Cyclone GR-2600, C14AB, C14AC, C14BA Series

T.C. NUMBER: T.C. 176

Models - GR-2600-	A2	A2A	A2B	B2	B5
New designation	726C14AB1, 2 332C14AB1, 2	709C14AC1, 2 579C14AC1, 2	--	585C14BA1, 2	586C14EA1, 2
Type - 14RA	16:9 reduction gears	16:9 reduction gears	16:9 reduction gears	16:9 or 16:7 reduction gears	16:9 or 16:7 reduction gears
Rating (with low impeller gear ratio):	7.0:1	7.0:1	7.0:1	7.03:1	6.96:1
Maximum continuous, hp, rpm, in.Hg., at:					
Rated pressure altitude (ft.)	1200-2100- 33.2-5400	1350-2300- 35.8-6200	1350-2300- 36.7-5800	1500-2400- 37.6-6700	--
Sea level pressure altitude	1200-2100- 35.0-S.L.	1350-2300- 37.5-S.L.	1350-2300- 39.0-S.L.	1500-2400- 39.7-S.L.	--
Take-off (five minutes), hp, rpm, in.Hg.	1550-2400-42.5	1600-2400-43.5	1600-2400-45.0	1700-2500-44.5	--
Rating (with high impeller gear ratio):	--	--	--	--	10.06:1
Maximum continuous, hp, rpm, in.Hg., at:					
Rated pressure altitude (ft.)	--	--	--	--	1350-2400-38.8-15000
Low critical pressure altitude (ft.)	--	--	--	--	1350-2400-43.0-9500
Fuel (minimum octane aviation gasoline - CFR Motor Method fuel)	95	--	90	95	--
Bore and stroke, in.	6.125 x 6.312	--	--	--	--
Displacement, cu. in.	2603	--	--	--	--
Compression ratio	7.1:1 (Model 332) 6.85:1 (Model 726)	7.1:1 (Model 709) 6.85:1 (Model 579)	6.3:1	6.9:1	6.9:1
Weight (dry), lbs.	1935, 1955	--	1935	1965, 1985	1980, 2000
Propeller shaft, SAE No.	50	--	--	--	--
Carburetion	Holley C.G. 1375 C,E,F,G,H,HAR or HE carburetor	Holley C.G. 1685 or F,H,HA,HAF, HAR or HB carburetor	Holley C.G. 1685 F,H,HA,HAR,HAF, HB or Stromberg PD12J1 or K1 carburetors	Holley C.G. 1685 F,H,HA,HAR,HB carburetor or Stromberg PT-13E2 carburetor	--
Ignition, dual	Scintilla SF14L- 3,4,SF14LN-3,4,9, or Bosch SF14LU-9 magnetos	--	--	Scintilla SF14L- 3,4,SF14LN-3,4, 9,10 or Bosch SF14LU-9, 10 magnetos	--
Ignition timing, degrees BIC	Both 20	--	--	--	--
NCIES	1, 2, 3, 4, 5	1, 3, 4, 5	1	1, 3, 4	1, 3, 4
Certification basis	Type Certificate No. 176				
Production basis	None. The manufacturer does not hold a production certificate for the production of engines under this type certificate and, therefore, each engine so produced is subject to a detailed inspection for workmanship and conformity with the approved data by a Civil Aeronautics Administration Agent. In addition, the engine must have a satisfactory run-in including 5 hours at rated power and speed. Upon satisfactory completion of the above, the agent will tag the engine with Tag Form ACA 186.				

NOTE 1. Maximum permissible cylinder head, barrel, and oil inlet temperatures, 450 degrees F., 325 degrees F., and 220 degrees F., respectively.

NOTE 2. This model includes previous GR-2600-A2 engines which have been modified to incorporate recent improvements.

NOTE 3. The new model designations, 332C14AB1, 709C14AC1, 585C14BA1, and 586C14EA1 will replace the GR-2600-A2, A2A, B2 and B5 designations respectively, in accordance with Wright Aeronautical Corporation's new engine designation system.

NOTE 4. Models C14AB2, AC2 and BA2 incorporate a torque meter.

NOTE 5. Models 332C14AB1, 2 and 709C14AC1, 2 also eligible as previously certificated with octane + .8 cc Tetra Ethyl lead per gallon of fuel.